ELEC4410 Control Systems Design Lecture 15: Observability

Julio H. Braslavsky

julio@ee.newcastle.edu.au

School of Electrical Engineering and Computer Science The University of Newcastle



Outline

- Observability
- Observability Gramian
- Duality Controllability-Observability
- Observability Tests
- Observation via Differentiation

The concept of observability is dual to that of controllability, and deals with the possibility of **estimating** the state of the system from the knowledge of its inputs and outputs.

Consider the LTI system

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u}, \quad \mathbf{A} \in \mathbb{R}^{n \times n}, \mathbf{B} \in \mathbb{R}^{n \times q}$$

$$\mathbf{y} = \mathbf{C}\mathbf{x} + \mathbf{D}\mathbf{u}, \quad \mathbf{C} \in \mathbb{R}^{p \times n}, \mathbf{D} \in \mathbb{R}^{p \times q}$$
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Observability: The state equation (SE), or the pair (A, C), is said to be observable if for any unknown initial state x(0), there exists a finite time $t_1 > 0$ such that the knowledge of the input u(t) and the output y(t) over $[0, t_1]$ suffices to determine uniquely the initial state x(0). Otherwise, the equation is said to be unobservable.

Example (Unobservable systems). The network shown in the figure below has two state variables: the current x_1 through the inductor and the voltage x_2 across the capacitor. The input u is a current source.



If u = 0, $x_2(0) = 0$ and $x_1(0) = a \neq 0$, then the output is identically zero. Any $x(0) = \begin{bmatrix} a \\ 0 \end{bmatrix}$ and $u(t) \equiv 0$ yield the same output $y(t) \equiv 0$.

Thus there is no way to uniquely determine the initial state $\begin{bmatrix} a \\ 0 \end{bmatrix}$ and the system is unobservable.

We have shown that the response of the state equation system is given by

$$\mathbf{y}(\mathbf{t}) = \mathbf{C}e^{\mathbf{A}\mathbf{t}}\mathbf{x}(\mathbf{0}) + \mathbf{C}\int_{\mathbf{0}}^{\mathbf{t}} e^{\mathbf{A}(\mathbf{t}-\mathbf{\tau})}\mathbf{B}\mathbf{U}(\mathbf{\tau})d\mathbf{\tau} + \mathbf{D}\mathbf{u}(\mathbf{t})$$

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$$Ce^{At}x(0) = \bar{y}(t), \tag{1}$$

where

$$\mathbf{\bar{y}}(t) = \mathbf{y}(t) - C \int_0^t e^{\mathbf{A}(t-\tau)} \mathbf{B} \mathbf{U}(\tau) d\tau - \mathbf{D} \mathbf{u}(t)$$

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is a known function. Thus the observability problem reduces to finding x(0) as the unique solution of (1).

For a fixed time t, Ce^{At} is a $p \times n$ real, constant matrix, and $\overline{y}(t)$ a constant $p \times 1$ vector.

Thus, in general, because p < n (there are less outputs than states) we cannot find a **unique** vector $x(\mathfrak{0})$ from

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Thus, in general, because p < n (there are less outputs than states) we cannot find a **unique** vector $x(\mathfrak{0})$ from

 $Ce^{At}x(0) = \bar{y}(t)$, for a given fixed t.

To determine x(0) uniquely we need to use the knowledge of y(t) and u(t) over a nonzero time interval.



Theorem (Gramian Observability Test). The state equation (SE) is observable if and only if the $n \times n$ matrix

$$W_{o}(t) = \int_{0}^{t} e^{A^{T}\tau} C^{T} C e^{A\tau} d\tau \qquad (WO)$$

is nonsingular for any t > 0.



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Note that observability only depends on the matrices A and C.



If the matrix A is Hurwitz (all eigenvalues have negative real part), then $W_o(t)$ converges for $t\to\infty,$ and we simply denote it by W_o ,

$$W_{o} = \int_{0}^{\infty} e^{\mathbf{A}^{\mathsf{T}} \boldsymbol{\tau}} \mathbf{C}^{\mathsf{T}} \mathbf{C} e^{\mathbf{A} \boldsymbol{\tau}} d\boldsymbol{\tau},$$

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The **Observability Gramian** can be computed by solving the linear matrix **Lyapunov equation**

$$W_{o}A + A^{\mathsf{T}}W_{o} = -C^{\mathsf{T}}C.$$



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In MATLAB the functions Ob = obsv(A,C) and Wo = gram(A',C')'respectively compute the observability matrix O and Gramian W_o . By checking the rank of O or W_o , we can determine if a pair (A, C) is observable.



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Proof. The pair (A, B) is controllable if and only if

$$W_{c}(t) = \int_{0}^{t} e^{A\tau} B B^{\mathsf{T}} e^{A^{\mathsf{T}}\tau} d\tau$$

is nonsingular for any t.

On the other hand, the pair (A^T, B^T) is observable if and only if, by replacing A by A^T and C by B^T in (WO),

$$W_{o}(t) = \int_{0}^{t} e^{A\tau} B B^{\mathsf{T}} e^{A^{\mathsf{T}}\tau} d\tau$$

is nonsingular for any t; the two conditions are thus identical.



The Duality between controllability and observability establishes that we can test the observability of a pair (A, C) by using the controllability tests that we already know on the pair (A^T, C^T) .

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Example. Consider the system

$$\dot{\mathbf{x}}(\mathbf{t}) = \begin{bmatrix} -2 & -2 & 0 \\ 0 & 0 & 1 \\ 0 & -3 & -4 \end{bmatrix} \mathbf{x}(\mathbf{t}) + \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ 1 & 1 \end{bmatrix} \mathbf{u}(\mathbf{t})$$
$$\mathbf{y}(\mathbf{t}) = \begin{bmatrix} 1 & 0 & 1 \end{bmatrix} \mathbf{x}(\mathbf{t})$$

By duality, we can check the observability of this system as the controllability of (A^T, C^T) ; for instance, via the matrix

$$\mathcal{C} = \begin{bmatrix} \mathbf{C}^{\mathsf{T}} \ \mathbf{A}^{\mathsf{T}} \mathbf{C}^{\mathsf{T}} \ \mathbf{A}^{\mathsf{T}\,2} \mathbf{C}^{\mathsf{T}} \end{bmatrix} = \begin{bmatrix} 1 & -2 & 4 \\ 0 & -5 & 16 \\ 1 & -4 & 11 \end{bmatrix}$$

which has rank 3, thus (A, C) is observable.



Test for Observability



Observability Tests

Theorem (Observability Tests). The following statements are equivalent.

- 1. The n-dimensional pair $(A,C),\,A\in \mathbb{R}^{n\times n},\,C\in \mathbb{R}^{p\times n},$ is observable.
- 2. The Observability Matrix

$$\mathfrak{O} = \begin{bmatrix} \mathbf{C} \\ \mathbf{C}\mathbf{A} \\ \mathbf{C}\mathbf{A}^2 \\ \vdots \\ \mathbf{C}\mathbf{A}^{n-1} \end{bmatrix}, \qquad \mathfrak{O} \in \mathbb{R}^{np \times n},$$

has rank n (full column rank).

3. The $n \times n$ matrix $W_o(t) = \int_0^t e^{A\tau} B B^T e^{A^T \tau} d\tau$ is non singular for all t > 0.





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From $\mathbf{\bar{y}}(t) = Ce^{At}x(0)$, since y(0) = Cx(0), $\mathbf{\bar{y}}(0) = CAx(0), \dots, \mathbf{\bar{y}}^{n-1}(0) = CA^{n-1}x(0)$, we have

$$\begin{bmatrix} C\\CA\\\cdots\\CA^{n-1}\end{bmatrix} x(0) = \begin{bmatrix} \ddot{y}(0)\\ \dot{\ddot{y}}(0)\\\cdots\\ \ddot{y}^{n-1}(0)\end{bmatrix} \quad \text{i.e.,} \quad \boxed{\Im x(0) = \vec{y}(0)}.$$

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$$\begin{bmatrix} \mathbf{C}\\ \mathbf{C}\mathbf{A}\\ \cdots\\ \mathbf{C}\mathbf{A}^{n-1} \end{bmatrix} \mathbf{x}(0) = \begin{bmatrix} \mathbf{\ddot{y}}(0)\\ \mathbf{\ddot{y}}(0)\\ \cdots\\ \mathbf{\ddot{y}}^{n-1}(0) \end{bmatrix} \quad \text{i.e.,} \quad \mathbf{O}\mathbf{x}(0) = \mathbf{\vec{y}}(0).$$

If the system is observable, then \mathfrak{O} is full column rank, and we know there exist a unique solution of $\mathfrak{O}x(\mathfrak{0}) = \vec{y}(\mathfrak{0})$ given by

$$\mathbf{x}(\mathbf{0}) = \left[\mathbf{O}^{\mathsf{T}}\mathbf{O}\right]^{-1}\mathbf{O}^{\mathsf{T}}\vec{\mathbf{y}}(\mathbf{0}).$$

Note that we still need to know $\mathbf{\bar{y}}(t)$ on a neighbourhood of t = 0 to be able to determine $\mathbf{\bar{y}}(0)$.

Is it practical to implement observation via differentiation? Although theoretically we could obtain x(0) by differentiation, in practice it is **not** recommended, since

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- on the other hand, **integration** "averages" high frequency noise, diminishing its effects in the computation of x(0).

It is much more convenient to implement observation by using integration, e.g., via the formula

$$\mathbf{x}(\mathbf{0}) = \mathbf{W}_{\mathbf{0}}^{-1}(\mathbf{t}_1) \int_{\mathbf{0}}^{\mathbf{t}_1} e^{\mathbf{A}^{\mathsf{T}} \mathbf{\tau}} \mathbf{C}^{\mathsf{T}} \mathbf{\bar{y}}(\mathbf{\tau}) d\mathbf{\tau}.$$

Examples



Example (Earth satellite).

A linearised state equation for a satellite in circular orbit is given by

$$\dot{\mathbf{x}}(\mathbf{t}) = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 3\omega_0^2 & 0 & 0 & 2\mathbf{r}_0 \omega_0 \\ 0 & 0 & 0 & 1 \\ 0 & -\frac{2\omega_0}{\mathbf{r}_0} & 0 & 0 \end{bmatrix} \mathbf{x}(\mathbf{t}) + \begin{bmatrix} 0 & 0 \\ 1 & 0 \\ 0 & 0 \\ 0 & \frac{1}{\mathbf{r}_0} \end{bmatrix} \mathbf{u}(\mathbf{t})$$
$$\mathbf{y}(\mathbf{t}) = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \mathbf{x}(\mathbf{t}) = \begin{bmatrix} \mathbf{r}(\mathbf{t}) \\ \theta(\mathbf{t}) \end{bmatrix}$$



where the first output is the (incremental) radial distance r(t) and the second the (incremental) angle $\theta(t)$.

The position of the satellite can be adjusted by means of the thrust forces $u_1(t)$ and $u_2(t)$. The nominal radius is r_0 and the nominal angular velocity ω_0 .

Example (continuation).

Suppose that only **radial** distance measurements

$$y_1(t) = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} x(t) = C_1 x(t)$$

are available on a specified time interval. The observability matrix in this case is

$$\begin{bmatrix} C_1 \\ AC_1 \\ A^2C_1 \\ A^3C_1 \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 3\omega_0^3 & 0 & 0 & 2r_0\omega_0 \\ 0 & -\omega_0^2 & 0 & 0 \end{bmatrix} \text{ which has rank 3.}$$

Therefore, radial measurement does **not** suffice to compute the complete orbit state.



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Therefore, radial measurement does **not** suffice to compute the complete orbit state.

On the other hand, measurement of angle,

$$y_1(t) = \begin{bmatrix} 0 & 0 & 1 & 0 \end{bmatrix} x(t) = C_2 x(t)$$

does suffice, as can be readily verified.

Example (Controllability and Observability of an RLC circuit). The RLC circuit below is modelled by the state equations

$$\begin{bmatrix} \dot{\mathbf{x}}_{1}(t) \\ \dot{\mathbf{x}}_{2}(t) \end{bmatrix} = \begin{bmatrix} -\frac{2}{RC} & \frac{1}{C} \\ -\frac{1}{L} & 0 \end{bmatrix} \begin{bmatrix} \mathbf{x}_{1}(t) \\ \mathbf{x}_{2}(t) \end{bmatrix} + \begin{bmatrix} \frac{1}{RC} \\ \frac{1}{L} \end{bmatrix} \mathbf{u}(t)$$
$$\mathbf{y}(t) = \begin{bmatrix} -1 & 0 \end{bmatrix} \begin{bmatrix} \mathbf{x}_{1}(t) \\ \mathbf{x}_{2}(t) \end{bmatrix} + \mathbf{u}(t)$$





Example (RLC circuit continuation). We test controllability by checking the rank of the Controllability Matrix,

$$\mathcal{C} = \begin{bmatrix} \mathbf{B} & \mathbf{A}\mathbf{B} \end{bmatrix} = \begin{bmatrix} \frac{1}{\mathbf{R}\mathbf{C}} & -\frac{2}{\mathbf{R}^2\mathbf{C}^2} + \frac{1}{\mathbf{L}\mathbf{C}} \\ \frac{1}{\mathbf{L}} & -\frac{1}{\mathbf{R}\mathbf{L}\mathbf{C}} \end{bmatrix}$$

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The rank of this matrix can be checked with the determinant,

$$\det \mathfrak{C} = \frac{1}{R^2 L C^2} - \frac{1}{L^2 C}$$

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$$\mathcal{C} = \begin{bmatrix} B & AB \end{bmatrix} = \begin{bmatrix} \frac{1}{RC} & -\frac{2}{R^2C^2} + \frac{1}{LC} \\ \frac{1}{L} & -\frac{1}{RLC} \end{bmatrix}$$

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$$\det \mathcal{C} = \frac{1}{R^2 L C^2} - \frac{1}{L^2 C}$$

The determinant is zero (and thus the system uncontrollable) if

$$\frac{1}{R^2 L C^2} - \frac{1}{L^2 C} = 0 \Leftrightarrow \boxed{R = \sqrt{\frac{L}{C}}}$$



Example (RLC circuit continuation). On the other hand, the Observability Matrix is

$$\mathfrak{O} = \begin{bmatrix} \mathbf{C} \\ \mathbf{C}\mathbf{A} \end{bmatrix} = \begin{bmatrix} -1 & \mathbf{0} \\ \frac{2}{\mathbf{R}\mathbf{C}} & -\frac{1}{\mathbf{C}} \end{bmatrix}$$

which is obviously full rank.

Hence the system is **always** observable, but becomes uncontrollable whenever $\mathbf{R} = \sqrt{L/C}$.



Example (RLC circuit continuation). Let's see what happens to the system transfer function when controllability is lost.

The calculation, using the known formula $G(s) = C(sI - A)^{-1}B + D \text{ gives}$

$$G(s) = \begin{bmatrix} -1 & 0 \end{bmatrix} \begin{bmatrix} s + \frac{2}{RC} & -\frac{1}{C} \\ \frac{1}{L} & s \end{bmatrix}^{-1} \begin{bmatrix} \frac{1}{RC} \\ \frac{1}{L} \end{bmatrix} + 1$$
$$= \frac{s\left(s + \frac{1}{RC}\right)}{s^2 + \frac{2}{RC}s + \frac{1}{LC}}$$



Example (RLC circuit continuation). The poles of the circuit transfer function are

$$s_{1,2} = -\frac{1}{RC} \pm \sqrt{\frac{1}{R^2C^2} - \frac{1}{LC}}.$$

Both roots have negative real part, and thus conclude that the system is **asymptotically stable** and **BIBO stable** for any value of **R**, **L** and **C**.

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In particular, for $R = \sqrt{L/C}$ (the value for which the system becomes uncontrollable), we have

$$s_{1,2} = -\frac{1}{RC} \pm \sqrt{\frac{1}{LC} - \frac{1}{LC}} = -\frac{1}{RC},$$

that is, the system has repeated roots, and

$$G(s) = \frac{s\left(s + \frac{1}{RC}\right)}{\left(s + \frac{1}{RC}\right)^2} = \frac{s}{\left(s + \frac{1}{RC}\right)}.$$



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A pole-zero cancellation reduces the system to first order.

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- Observability depends o the matrices A and C of the state equation of the system. The pair (A, C) is observable if and only if

$$\operatorname{rank} \mathfrak{O} = \operatorname{rank} \begin{bmatrix} c \\ cA \\ \vdots \\ cA^{n-1} \end{bmatrix} = n$$



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As for Controllability, Observability is invariant with respect to change of coordinates (algebraic equivalence transformations).